

# Design of GRACE-like Small Satellite Constellations for Improved Temporal Gravity Measurements

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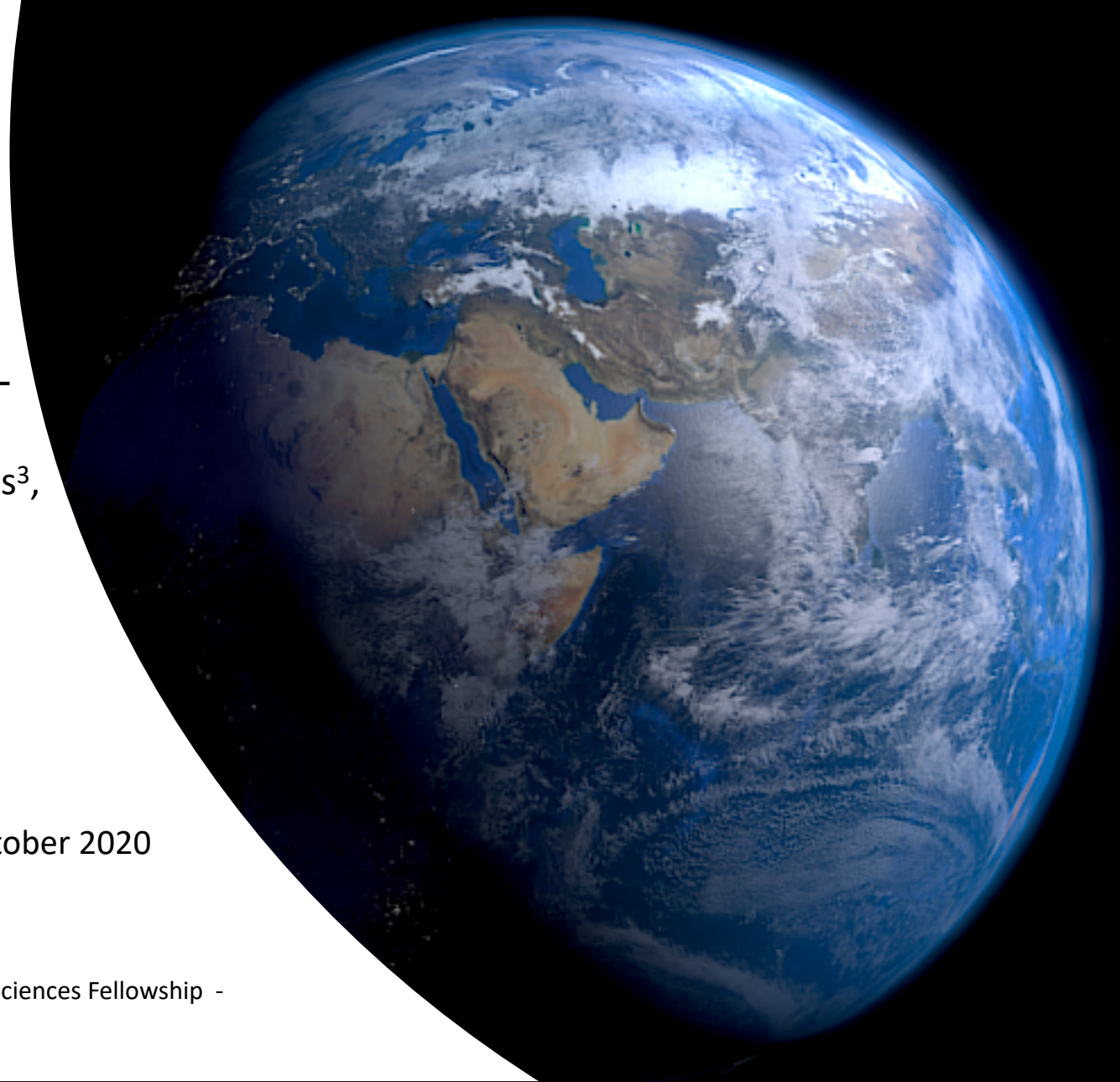
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<sup>2</sup>Jet Propulsion Laboratory, California Institute of Technology

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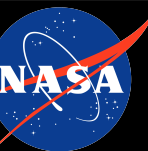
GRACE/GRACE-FO SCIENCE TEAM MEETING 2020 , 26-29 October 2020

This work is supported by NASA Headquarters under the NASA Earth and Space Sciences Fellowship -  
Grant: #18-EARTH18F-0380.



# Outline

- Motivation
- Method
  - Numerical search algorithm
  - NSGA-II
  - Objective function
- Preliminary analysis
  - Setup
  - One day solutions
- Next steps & end goal



# Outline

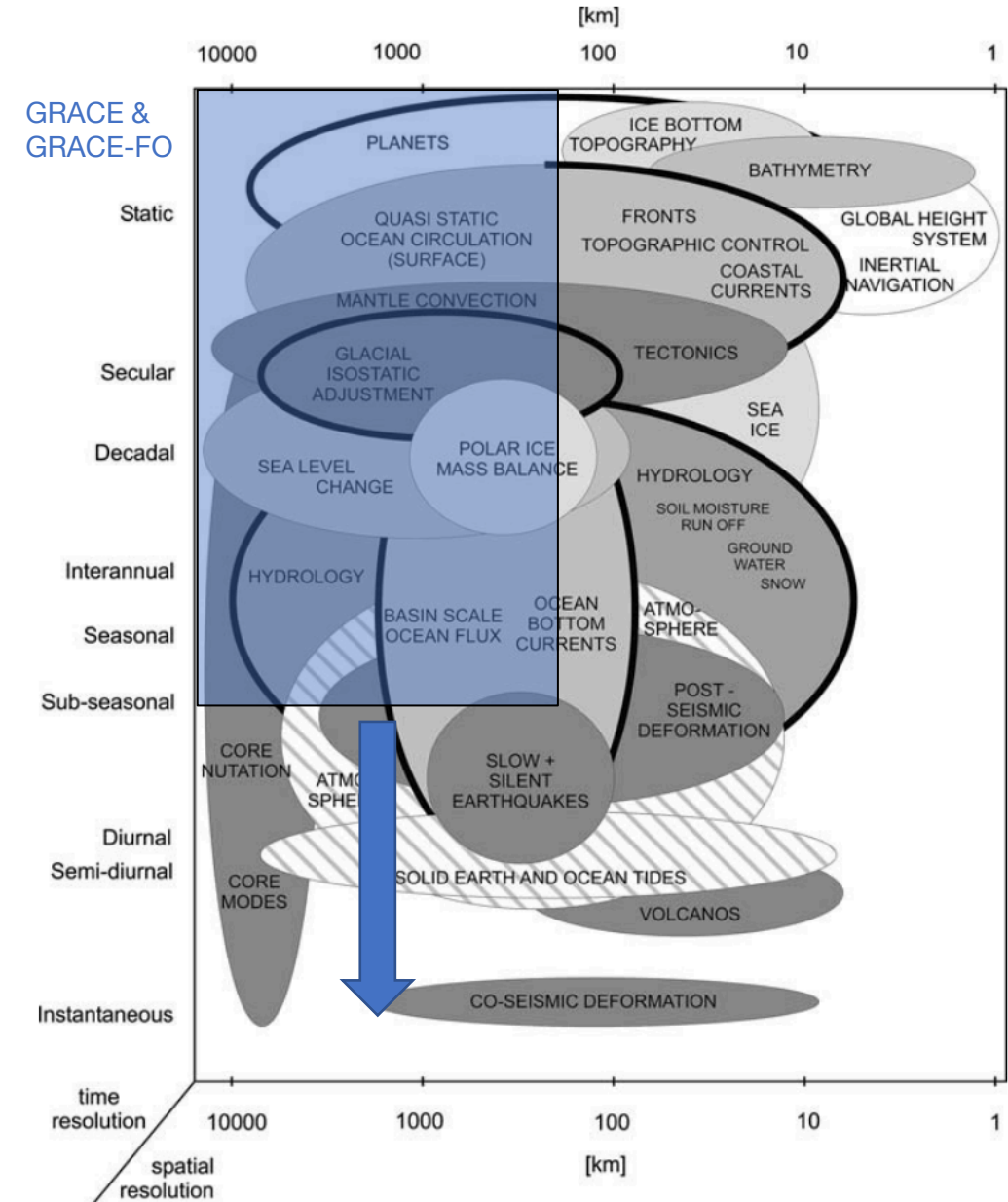
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# Beyond GRACE-FO

## 2016-2020

LIST	Land surface topography for landslide hazards and water runoff	LEO, SSO	Laser altimeter	300
PATH	High-frequency, all-weather temperature and humidity soundings for weather forecasting and sea-surface temperature <sup>a</sup>	GEO	Microwave array spectrometer	450
GRACE-II	High-temporal-resolution gravity fields for tracking large-scale water movement	LEO, SSO	Microwave or laser ranging system	450
SCLP	Snow accumulation for freshwater availability	LEO, SSO	Ku- and X-band radars K- and Ka-band radiometers	500
GACM	Ozone and related gases for intercontinental air quality and stratospheric ozone layer prediction	LEO, SSO	UV spectrometer IR spectrometer Microwave limb sounder	600
3D-Winds (Demo)	Tropospheric winds for weather forecasting and pollution transport	LEO, SSO	Doppler lidar	650

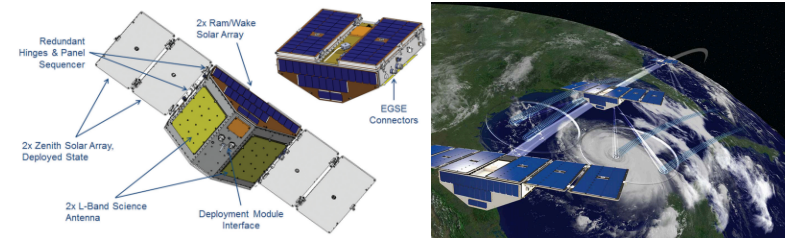


Adapted from: Sneeuw et al. 2005



# Why Smallsats?

- Due to
  - Miniaturization of instruments
  - Increasing smallsats capabilities
  - Insensitive to single point failures
- Smallsats are beginning to be used for scientific missions
  - Antarctic Glacier and Sea Ice Observation with a 3U Cube Satellite (Wu et al., 2017)
    - Launched: Sept 25, 2015
  - CYGNSS : 8 Smallsats constellation (27.5 kg) measure ocean surface wind speed for hurricane forecasting (Ru et al., 2013)
    - Launched: Dec 15, 2016
- Smallsats could be **low-cost alternatives** to large missions
  - A GRACE-like smallsat production cost could be \$5M–\$10M for each satellite unit (T. Yunk, 2020)
- Can we use a smallsat constellation if they are good enough?



Source: CYGNSS Handbook/NASA

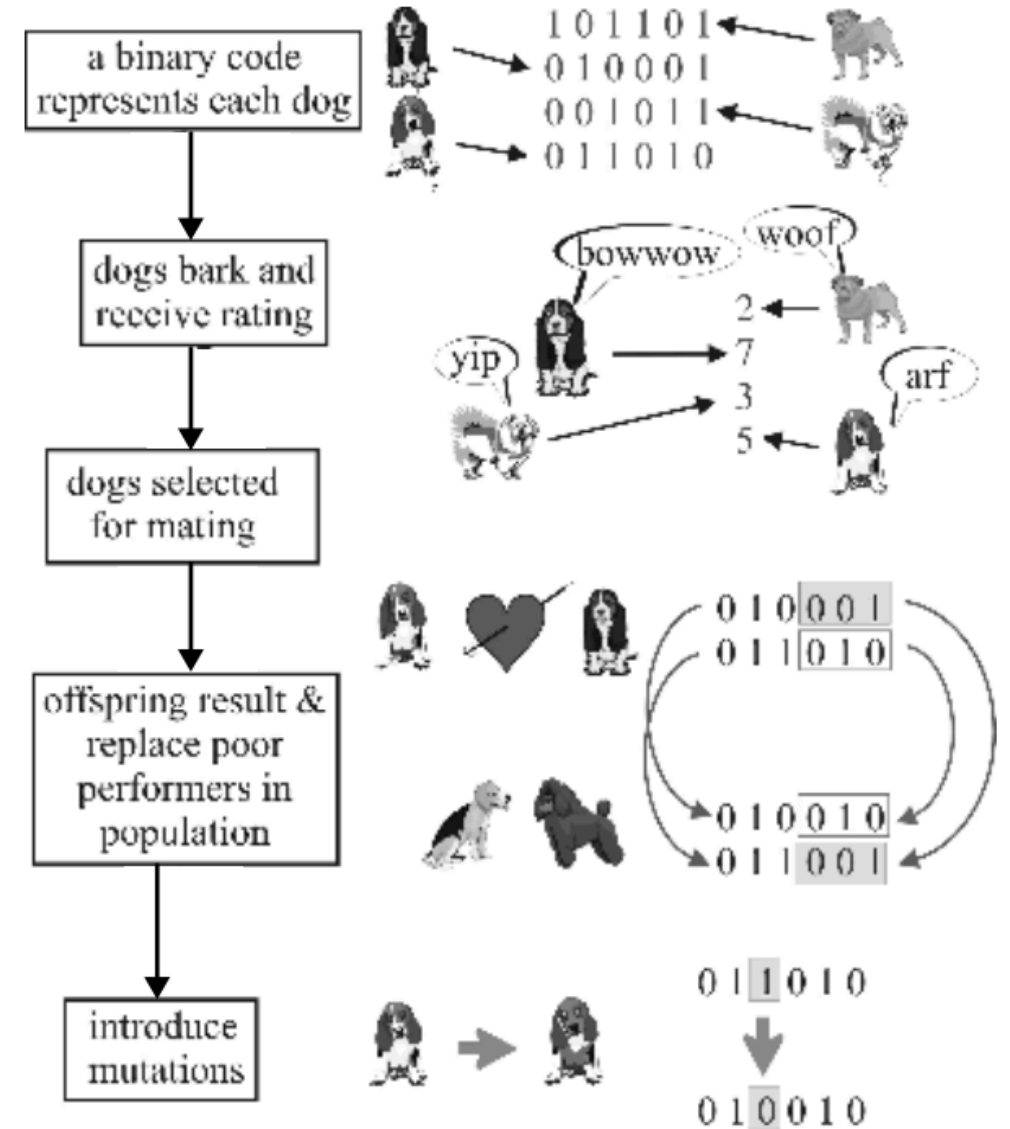
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# Genetic algorithm

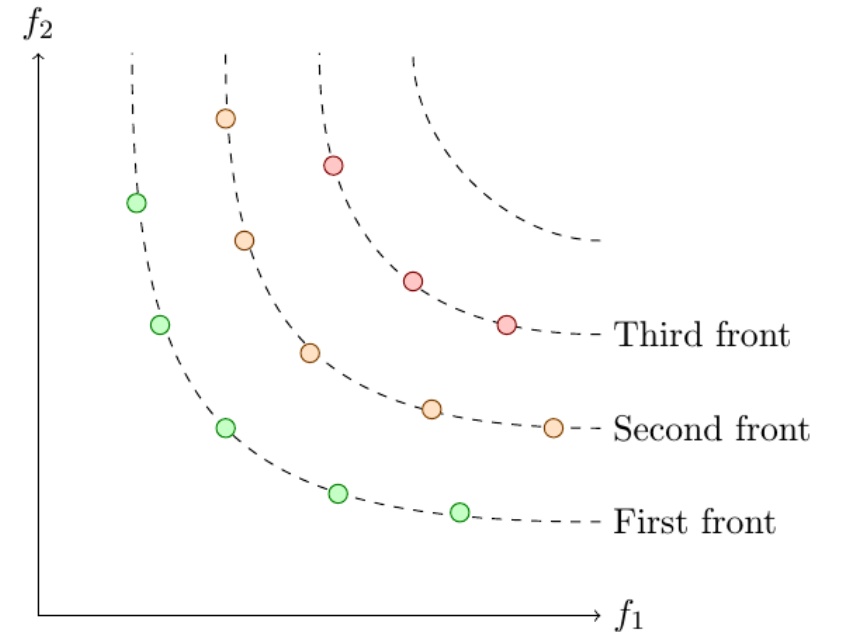
- Proven technique for large scale constellation design (Stern et al., 2018, Savitri et al., 2017, Iran-Pour et al., 2014)
- Advantage
  - Converges “fast” for nonlinear problems to the **vicinity** of the solutions
- Drawback
  - Does not converge fast to the final solution



Source: Haupt and Haupt 2004

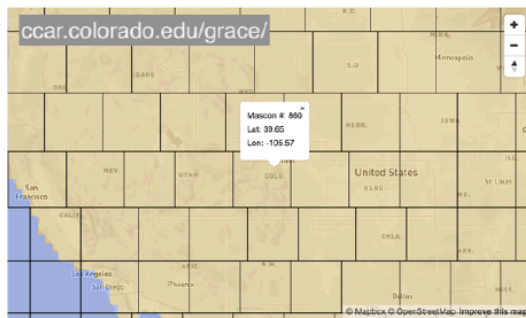
# Nondominated Sorting Genetic Algorithm (NSGA-II)

- Multi-objective GA
- Fast nondominated sorting
  - Pareto fronts
- Diversity through a crowded-comparison operator
- Repeated iterations lead to a global optimum

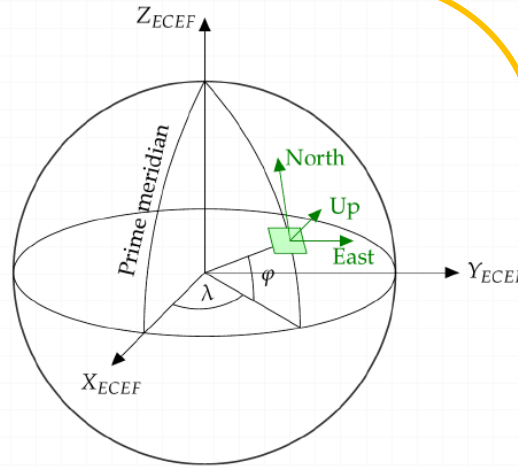


# Objective function

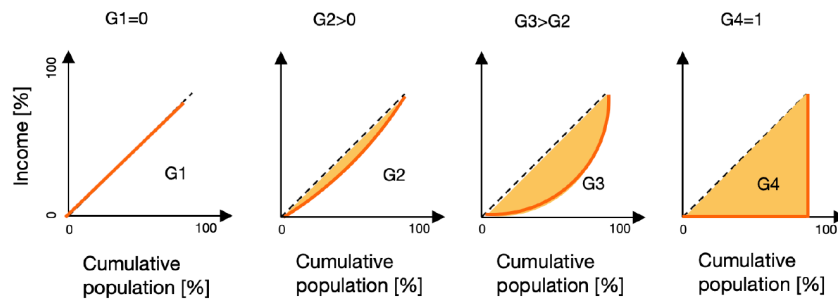
Spatial objective function ( $J_{sc}$ )



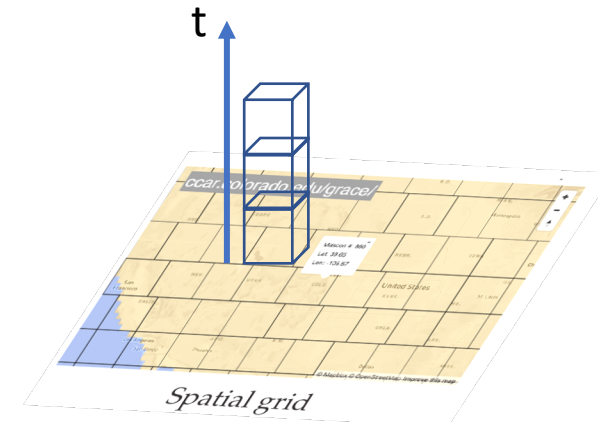
Spatial grid



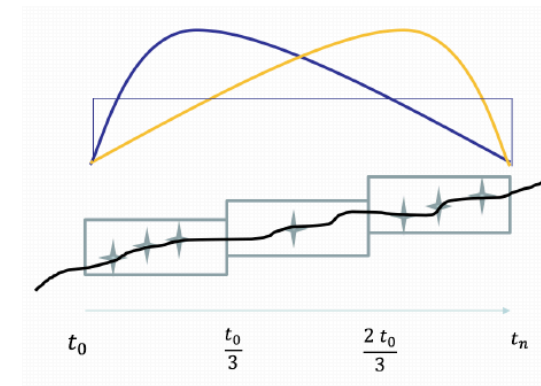
N-S vs E-W information



Uniformity

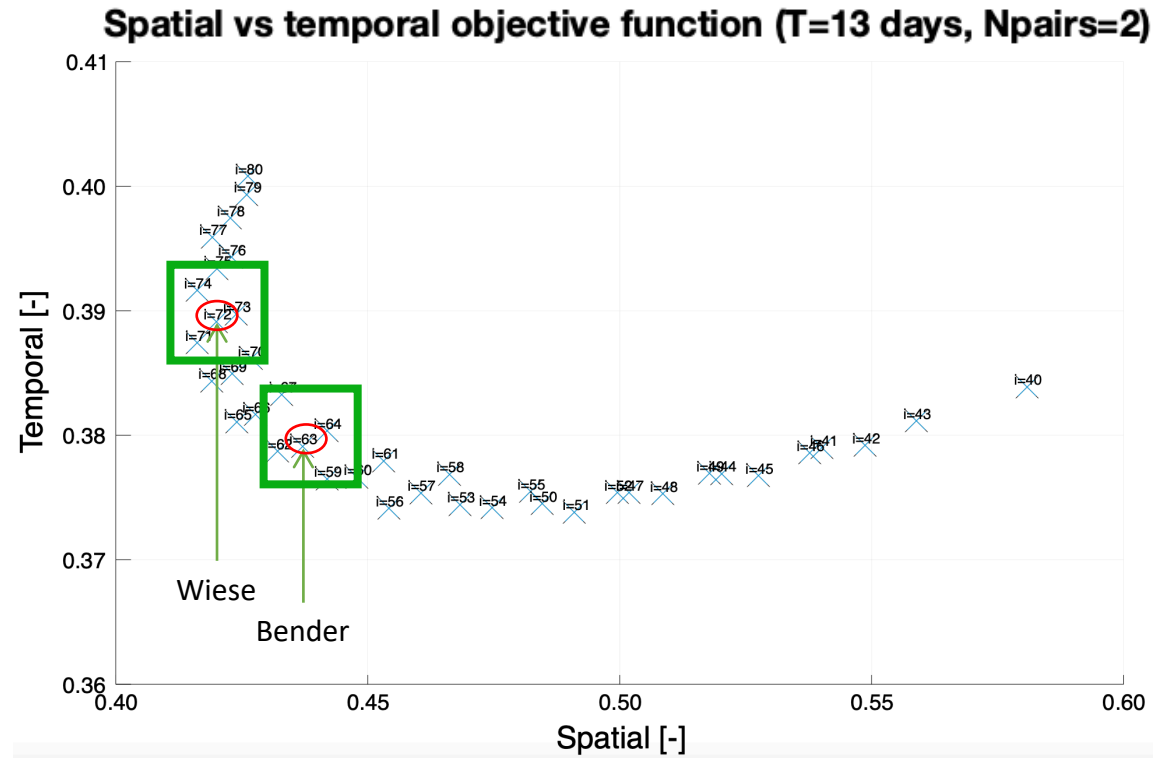


Temporal objective function ( $J_{TC}$ )



Temporal grid

# Selecting the desired constellation

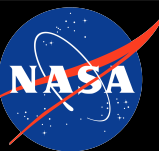


Simulations from the literature have shown what the area of interest is.



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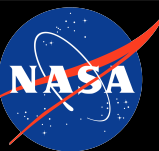


# GA variables

- Solution period
  - 1 day
- First satellite pair
  - $i_1=90$
  - $M_1=5$
  - $\Omega_1=5$
- Altitude
  - $h \approx 500(5)$  km

Fixed parameters

- Search space
  - $n_{\text{pairs}} = [6 \ 9 \ 12 \ 24]$
  - $i_{n>1} = [40 - 89, 91 - 140]$ 
    - intervals of 3.26 deg
    - $2 \cdot 2^4$  options
  - $M_{n>1} = [5 - 360]$ 
    - intervals of 50 deg
    - $2^3$  options
  - $\Omega_{n>1} = [5 - 360]$ 
    - intervals of 50 deg
    - $2^3$  options
- **Discrete** analysis of the search space



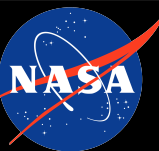
# GEODYN setup (Truth = Nominal)

- Truth: EIGEN-GL04C + EMCWF + OMCT
- Nominal: EIGEN-GL04C + EMCWF + OMCT



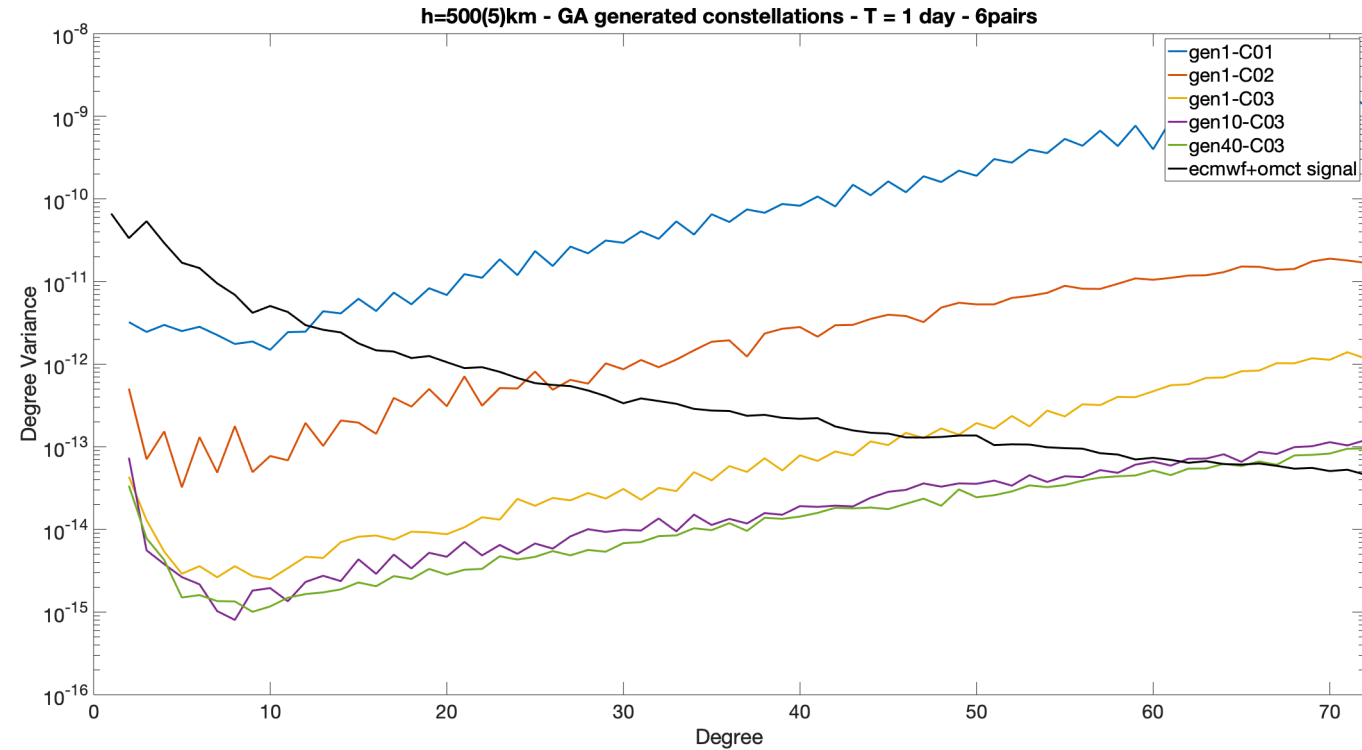
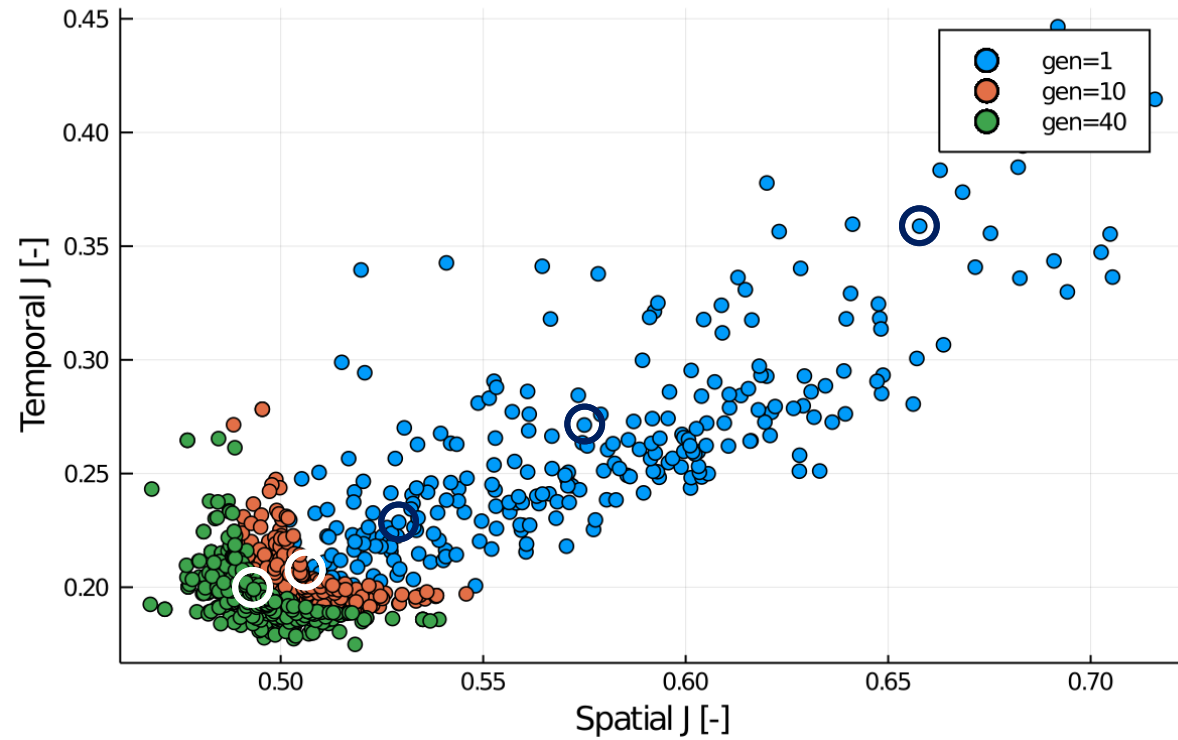
- Setup deck and initiation script includes:
  - Tidal aliasing (FES2004 - GOT.00)
  - AOD errors (EMCWF + NCEP) – (OMCT + MOG2D)
  - Range rate errors (GRACE-FO levels)
  - Accelerometer errors (GRACE-FO levels)
  - Satellite position errors (10 cm)

➤ Recovered CS error due only to Range rate + Accelerometer Error

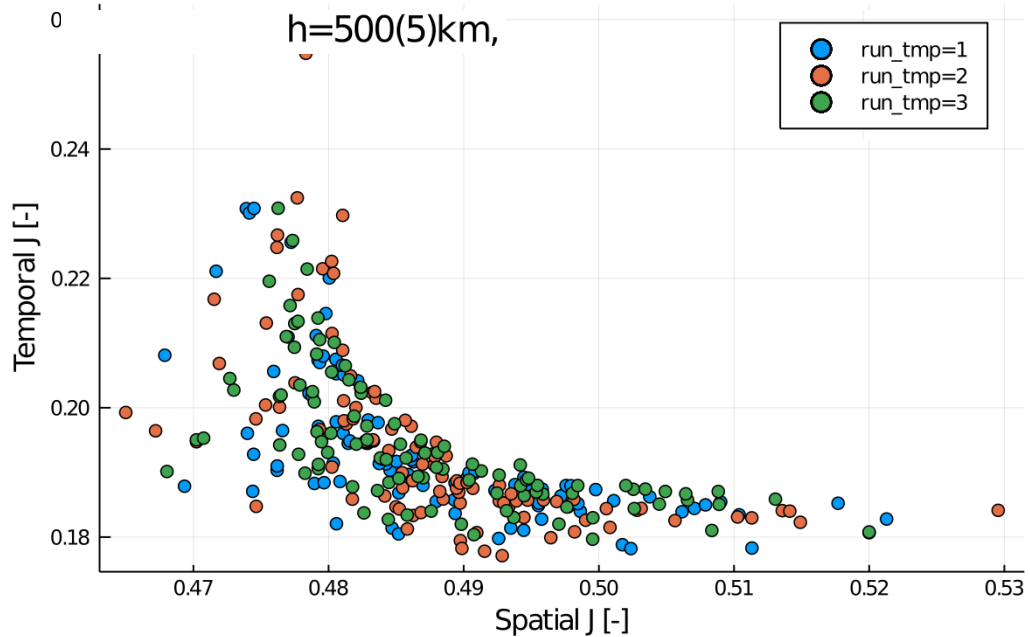


# Generational evolution

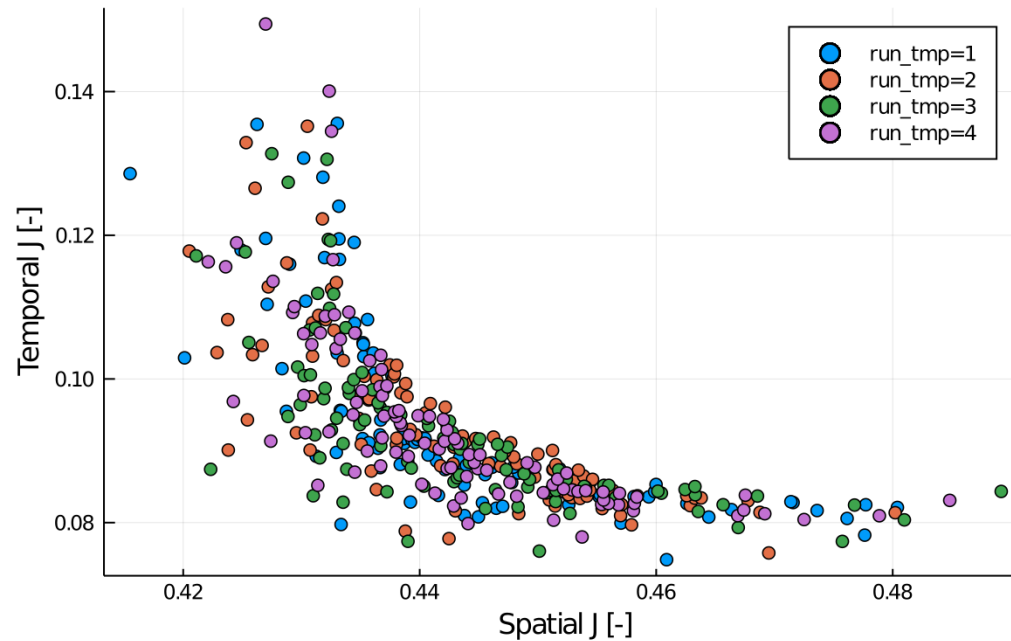
$h=500(5)km, T_{opt}=1d, N_{pair}=6$



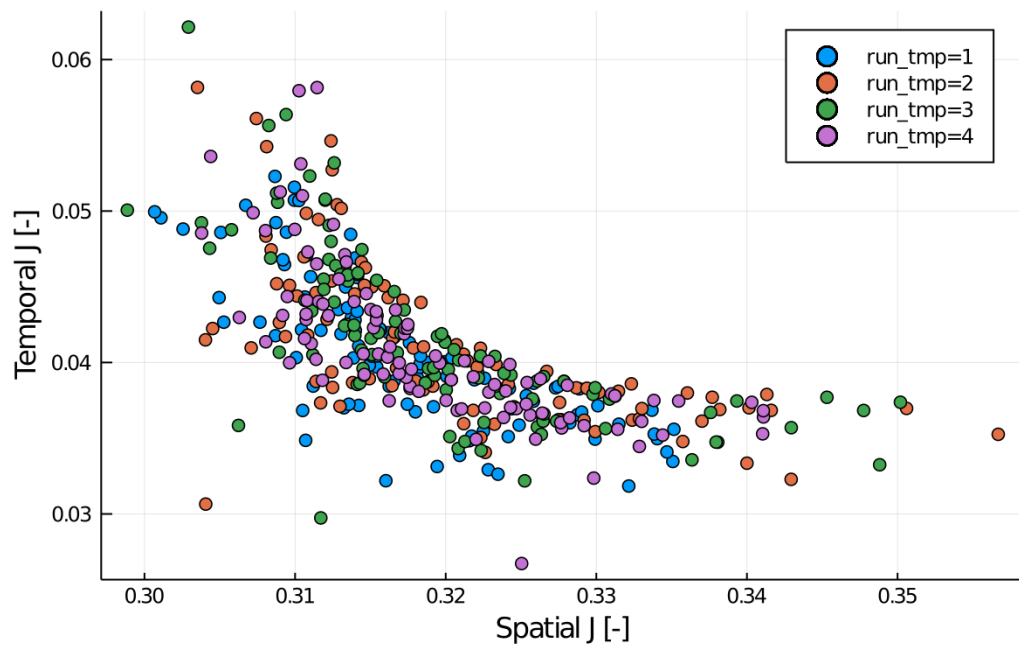
Topt=1d, Npairs=6



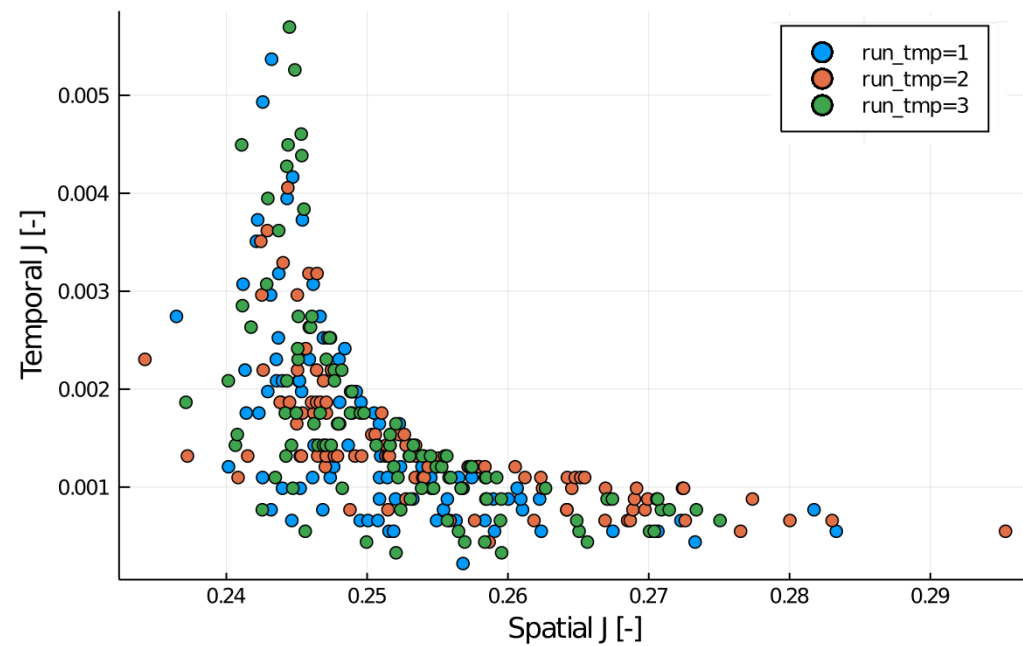
h=500(5)km, Topt=1d, Npairs=9



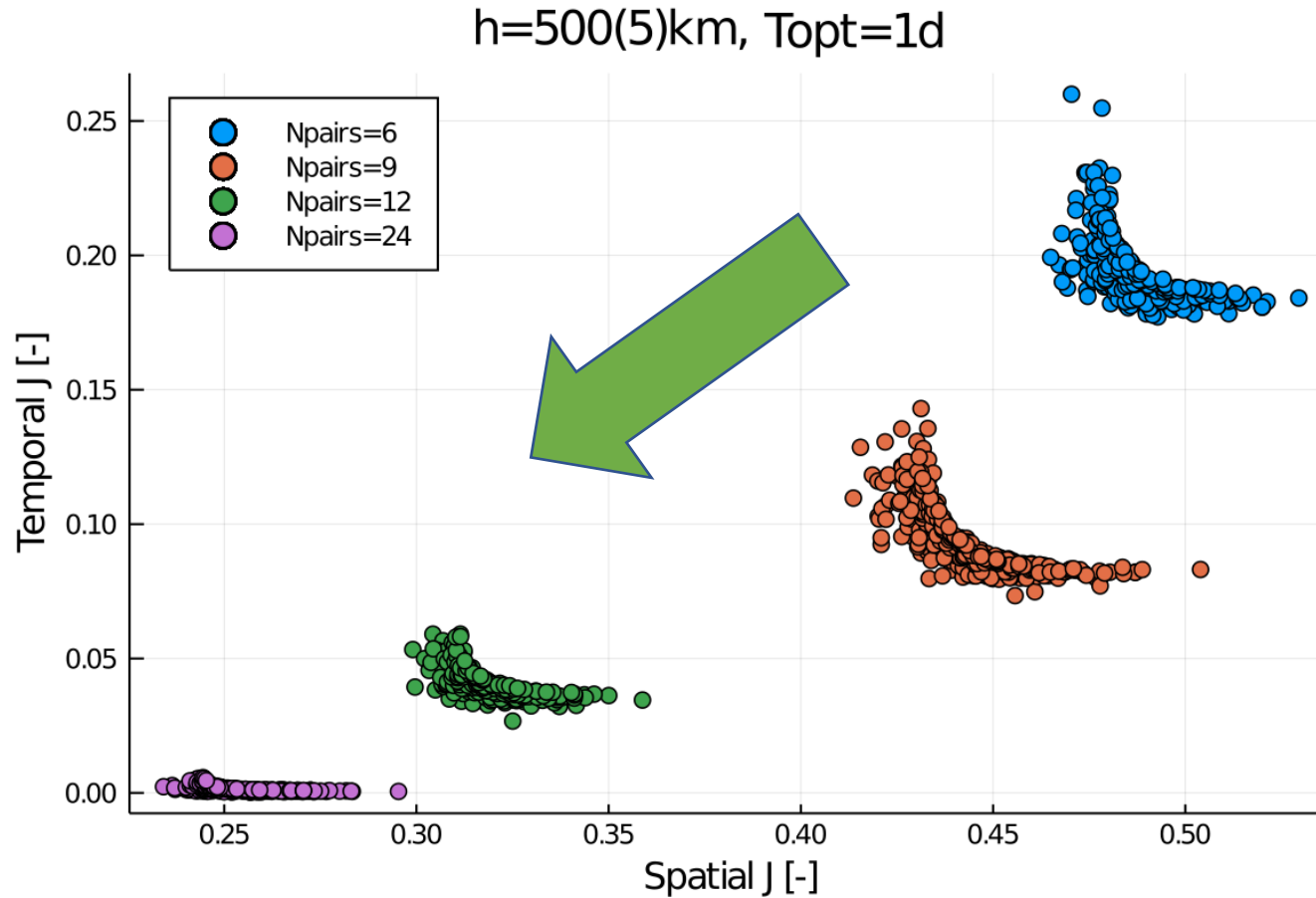
h=500(5)km, Topt=1d, Npairs=12



h=500(5)km, Topt=1d, Npairs=24

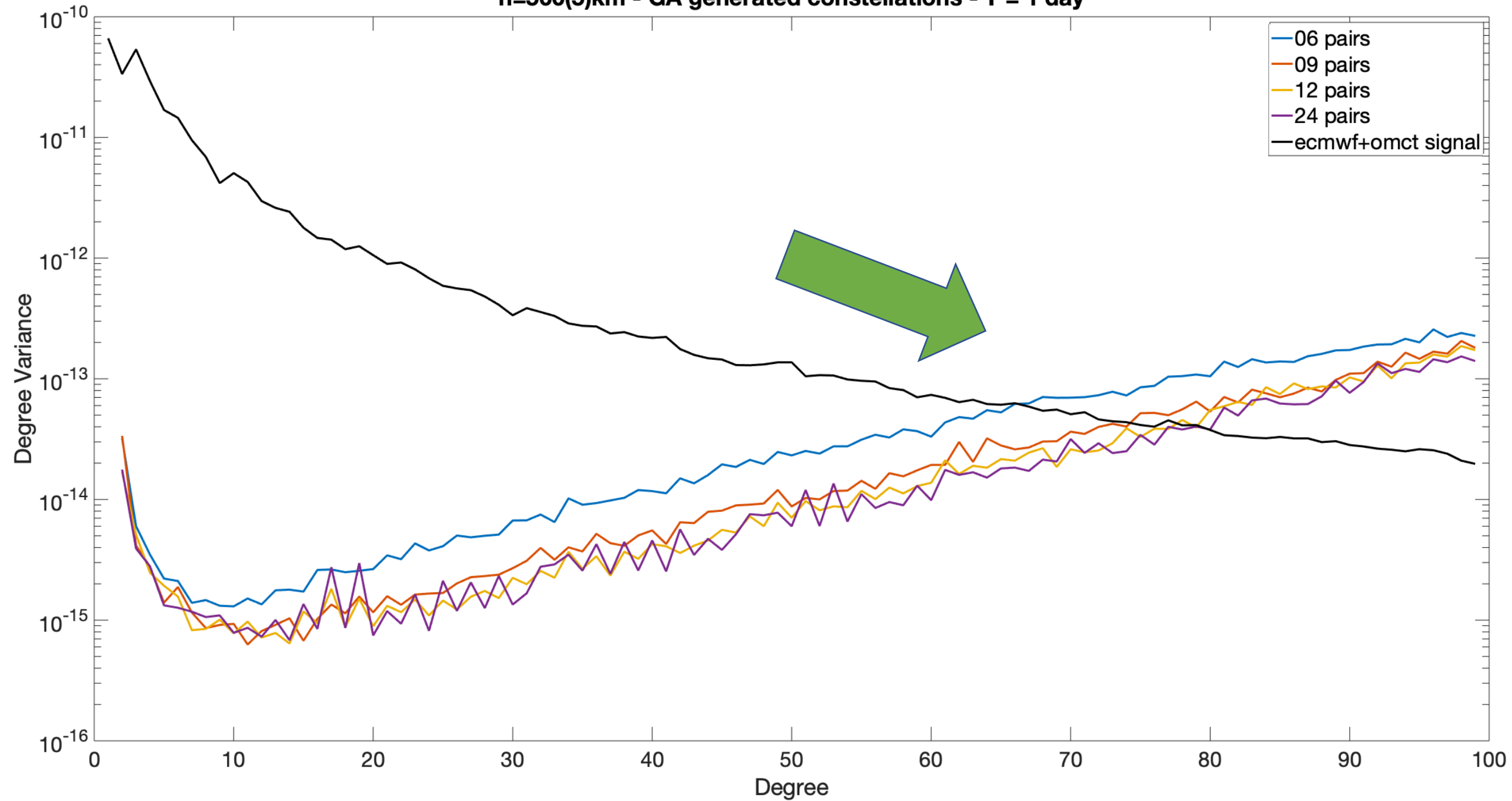


# Diminishing return

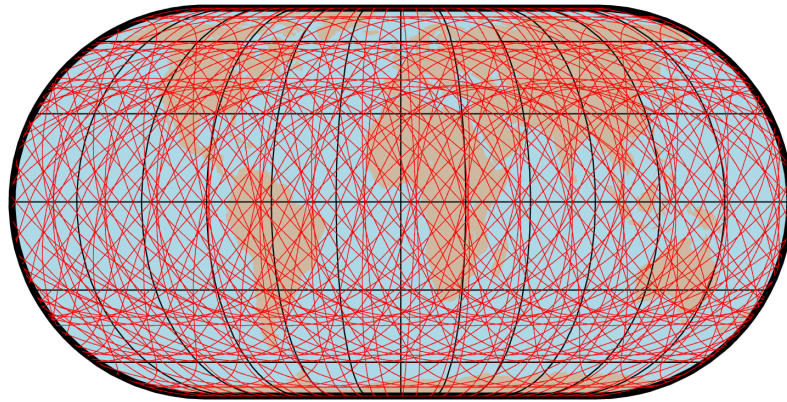




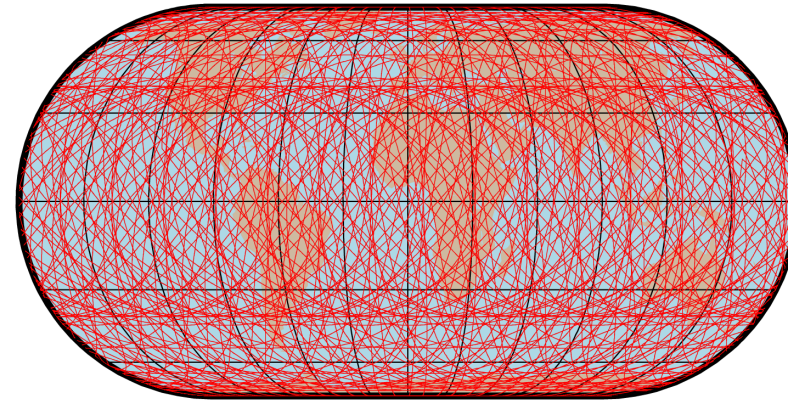
**h=500(5)km - GA generated constellations - T = 1 day**



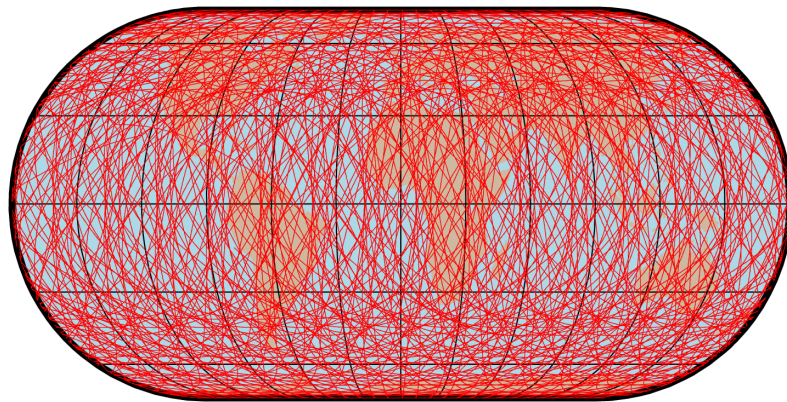
# Ground tracks



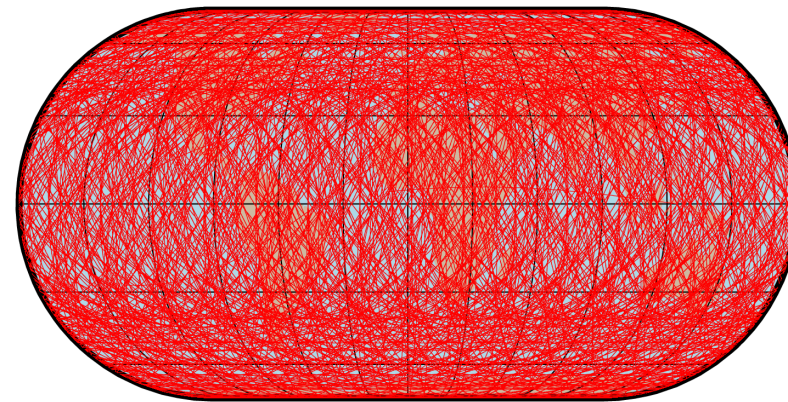
6 sat



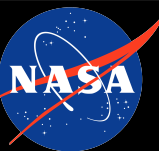
9 sat



12 sat



24 sat

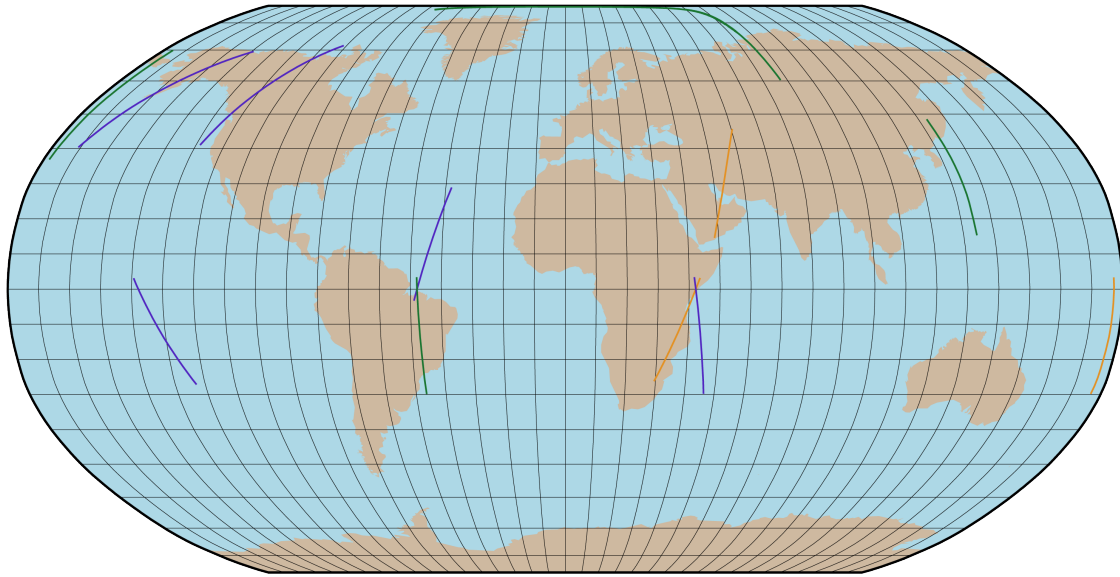


# GA solution - 6, 9 pairs

Pair	i [deg]	M [deg]	RAAN [deg]	Type [-]
1	90.0	5	5	polar
2	100.8	90	270	retrograde
3	113.9	270	180	retrograde
4	126.93	90	270	retrograde
5	130.2	180	180	retrograde
6	136.73	360	270	retrograde

Pair	i [deg]	M [deg]	RAAN [deg]	Type [-]
1	90	5	5	polar
2	89	360	315	polar
3	91	45	270	polar
4	46.53	45	270	prograde
5	107.33	225	315	retrograde
6	110.6	225	135	retrograde
7	123.67	45	45	retrograde
8	130.2	90	90	retrograde
9	136.73	135	315	retrograde

# GA solution - 12 pairs



Pair	i [deg]	M [deg]	RAAN [deg]	Type [-]
1	90.0	5	5	polar
2	91.0	270	180	polar
3	91.0	315	180	polar
4	94.3	360	135	polar
5	43.3	135	45	prograde
6	53.1	135	135	prograde
7	79.2	90	135	prograde
8	113.9	135	135	retrograde
9	123.7	315	315	retrograde
10	123.7	315	45	retrograde
11	126.9	180	135	retrograde
12	140.0	90	180	retrograde



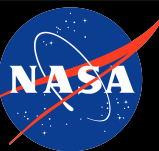
# GA solution - 24 pairs

Pair	i [deg]	M [deg]	RAAN [deg]	Type [-]
1	90.0	5	5	polar
2	91.0	225	180	polar
3	91.0	315	135	polar
4	91.0	315	270	polar
5	40.0	135	135	prograde
6	43.3	225	180	prograde
7	43.3	360	45	prograde
8	56.3	270	135	prograde
9	59.6	315	270	prograde
10	72.7	270	180	prograde
11	72.7	315	90	prograde
12	75.9	45	135	prograde

Pair	i [deg]	M [deg]	RAAN [deg]	Type [-]
13	75.9	180	135	prograde
14	79.2	315	180	prograde
15	85.7	270	90	prograde
16	104.0	180	315	retrograde
17	107.3	225	135	retrograde
18	113.9	225	225	retrograde
19	113.9	315	90	retrograde
20	113.9	360	45	retrograde
21	126.9	135	225	retrograde
22	130.2	180	180	retrograde
23	136.7	270	45	retrograde
24	140.0	225	315	retrograde

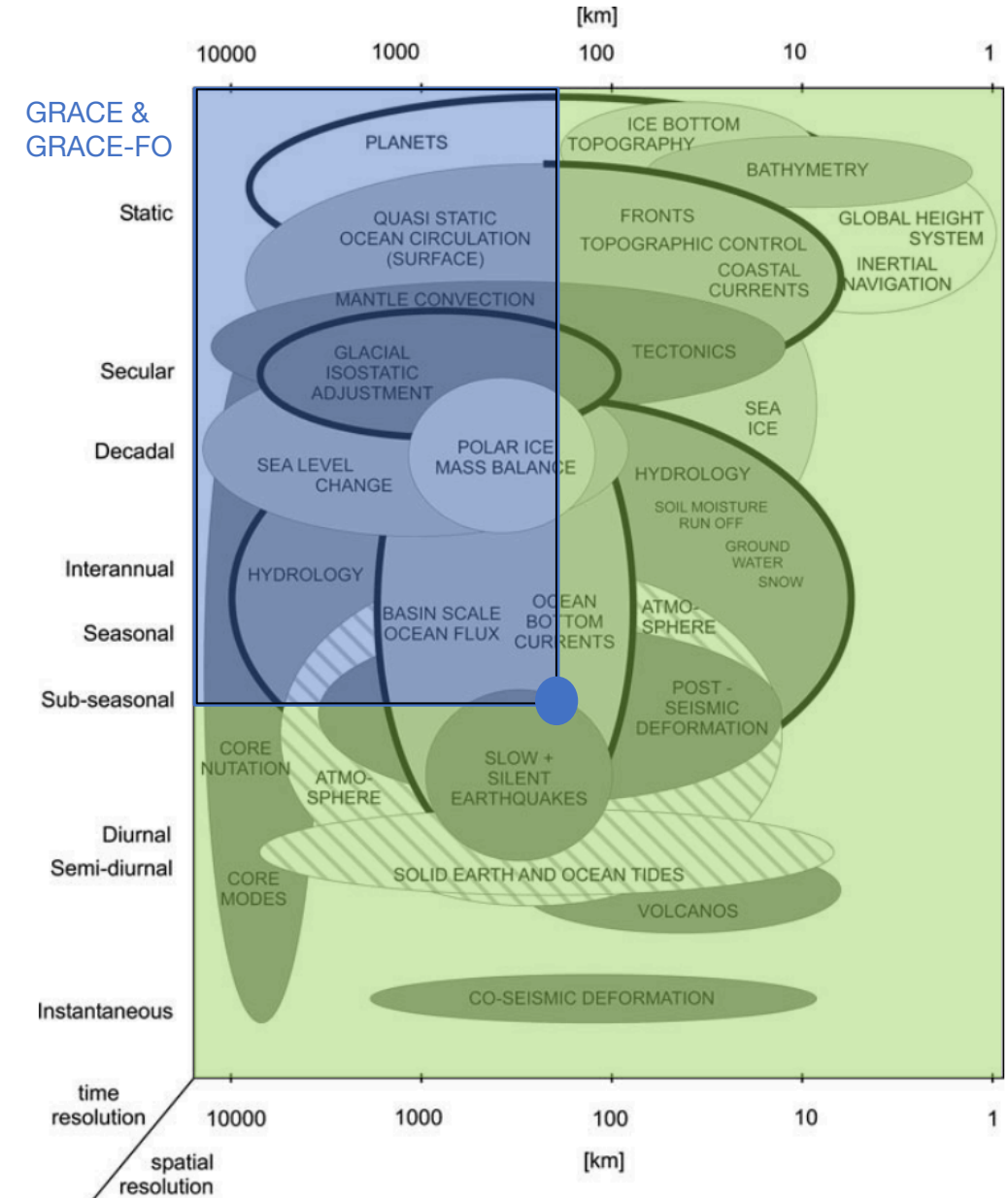
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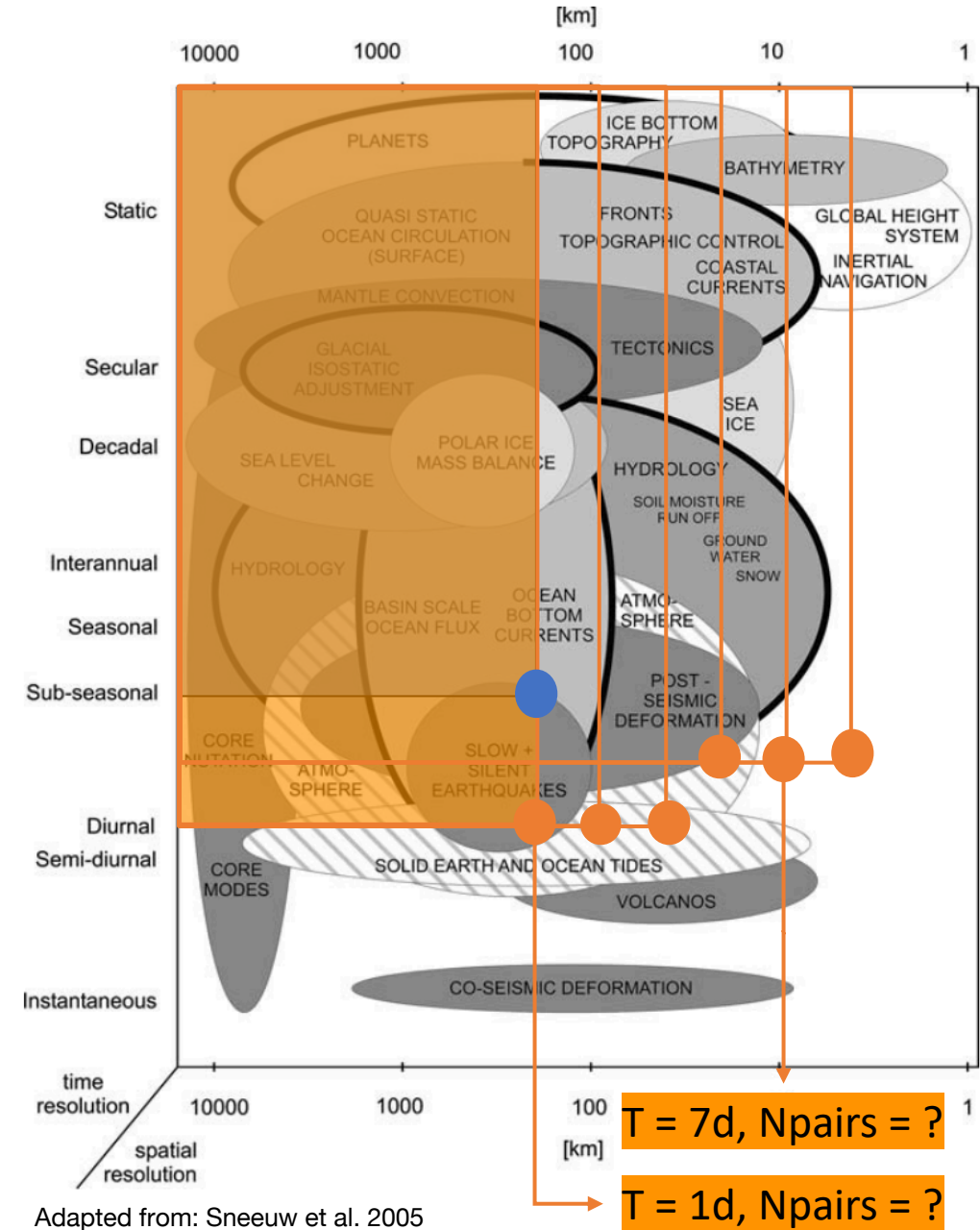
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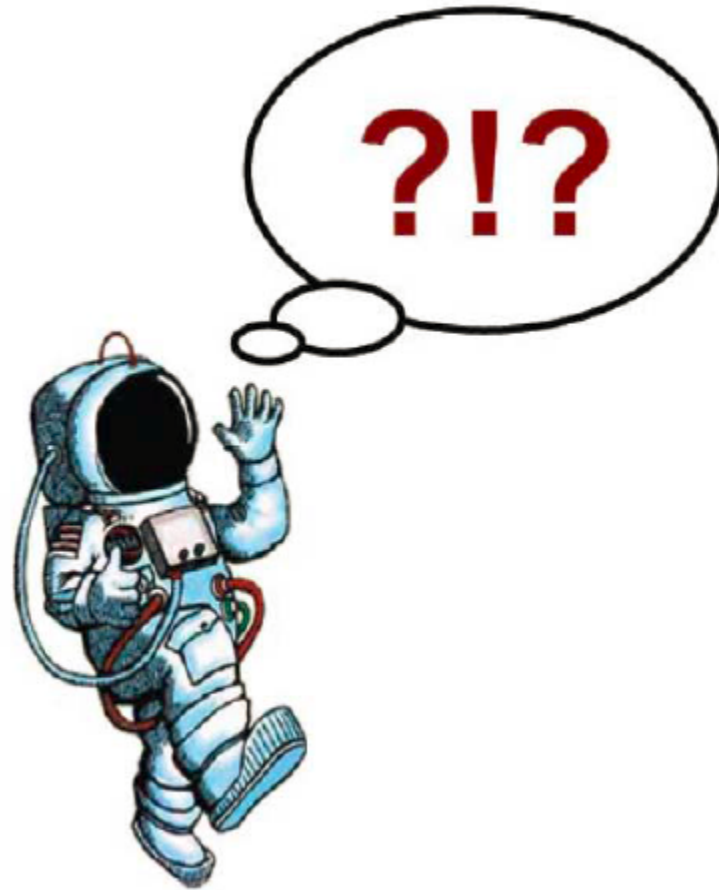
Adapted from: Sneeuw et al. 2005

# Next steps & end goal

- Relate
  - Given: Temporal resolution
  - Npairs required (& orbital config)
  - Spatial resolution achievable
  - To which level of accuracy
- Goal
  - Sub daily solutions
  - Significantly reduce temporal aliasing



Adapted from: Sneeuw et al. 2005



Source: TU Delft

